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Amplitudes of vibration for a parametrically excited inclined cable close to two-to-one internal resonance

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Abstract

This paper presents a study of how different vibration modes contribute to the dynamics of an inclined cable that is parametrically excited close to a 2 : 1 internal resonance. The behaviour of inclined cables is important for the design and analysis of cable-stay bridges. In this work the cable vibrations are modelled by a four-mode model. This type of model has been used previously to study the onset of cable sway motion caused by internal resonances which occur due to the non-linear modal coupling terms. A bifurcation study is carried out with numerical continuation techniques applied to the scaled and averaged modal equations. As part of this analysis, the amplitudes of the cable vibration response to support inputs is computed. These theoretical results are compared with experimental measurements taken from a 5.4 m long inclined cable with a vertical support input at the lower end. In general this comparison shows a very high level of agreement.

keywords:
Cable vibration, internal resonance, sway motion, modal interaction, bifurcation analysis.


1 Introduction

Inclined cables are used to support the bridge deck in cable-stay bridges. The cables are typically lightly damped, and when the bridge deck oscillates it provides a support motion input to the cable. This type of excitation can lead to large amplitude vibrations of the cable. A case of particular interest is when an internal resonance occurs between the in-plane and out-of-plane modes of vibration of the cable. This phenomenon was studied in [1] by using nonlinear Mathieu-type equations to model parametric resonance between the in-plane and out-of-plane modes of vibration of the cable [2]. The most significant resonance occurs when the associated modal frequencies of the second in-plane and first out-of-plane mode are at a ratio of 2 : 1. Close to resonance, relatively small deck inputs are sufficient to trigger the out-of-plane sway motion of the cable. This onset of sway motion can be formulated as a stability problem. In this case, zero sway motion is considered a stable situation, and the onset of sway motion can be interpreted as the loss of stability of the zero solution. This approach has been considered by many authors; see for example [3, 4, 5, 6, 7, 8, 9, 10, 11, 12] and references therein. The general formulations for cable dynamics are discussed in [13, 14].

In this paper the Warnitchai equations [15] (see also [16] for a detailed derivation) are used to model the vibration of the cable. It is assumed that the longitudinal vibrations of the cable can be neglected, so that the planes of interest are vertical (in-plane) and sway (out-of-plane). Four modes are included in the model, two in-plane and two out-of-plane, which enables the important low frequency dynamic behaviour to be modelled [7, 8]. The Warnitchai equations are scaled and averaged using the same procedure as previously presented in [11, 12]. Then a bifurcation study is carried by means of numerical continuation [17] with the software package AUTO [18]. In this way, the stability boundaries for the zero-sway solution can be computed in the form of a series of resonance (or Arnold) tongues. In fact, the discussion will be limited to the 2 : 1 resonance tongue; additional resonances are discussed in [12].

In addition to computing the resonance tongues, amplitude information can also be obtained. This type of amplitude information has not typically been discussed in detail or experimentally verified in previous work. In order to compare the amplitude information from the model with experimental results, tests were carried out on an inclined 5.4 m long cable [19]. The modal data from the model was transformed into displacement values for the mid
and quarter points of the experimental cable. When compared, the simulation and experimental results show a high level of agreement for most part of the curves [20]. A discrepancy between simulated and experimental results occurs close to tip of the resonance tongue where more complex dynamic effects are expected to occur. Such effects are not fully captured by the four mode model, but can be explained using the results of the bifurcation study.

The remainder of this paper is structured as follows. In Sec. 2 the equations of motion for the inclined cable model are presented as derived via scaling and first-order averaging. Section 3 describes the experimental setup and tests conducted in the laboratory. Section 4 describes the bifurcation study and numerical continuation analysis, as well as a comparison with experimental measurements. Conclusions are drawn in Sec. 5.

2 The theoretical model

The modal equations of motion for an inclined taut cable derived by Warnitchai et al. [15] are used here as the basis for the theoretical investigation. These modal equations are then scaled and averaged as described in [11, 12] (see also [16] for a description of the averaging technique) to give equations relating to the amplitude of response of each mode. The averaged equations are the basis for the bifurcation study described in section 4. For completeness, we set out the main steps of this derivation.

A schematic of the cable is shown in Fig. 1, where $v$ and $w$ are out-of-plane (sway) and in-plane displacements of the cable, respectively, and $\theta$ is the angle of inclination measured from the horizontal line in the gravity plane. Axial vibrations of the cable are neglected in this model since they occur at much higher frequencies.

The cable is rigidly supported at the upper end and a vertical support input is included at the lower end.

2.1 Modal equations

In the Warnitchai et al. [15] derivation, the dynamic response of the cable is split into a quasi-static motion (which ensures that the moving boundary condition at the lower support is met by considering the movement of a
mass-less tendon between the supports) and a modal response

\[ v_d(x, t) = v_m(x, t) = \sum_{n=1}^{\infty} \phi_n(x)y_n(t), \]
\[ w_d(x, t) = w_q(x, t) + w_m(x, t) = w_q(x, t) + \sum_{n=1}^{\infty} \psi_n(x)z_n(t), \]  

where the subscripts \(d\), \(q\) and \(m\) relate to dynamic, quasi-static and modal displacements, respectively, the spatial functions \(\phi(x)\) and \(\psi(x)\) are the in-plane and out-of-plane linear modeshapes of a cable with fixed ends, and \(y_n(t)\) and \(z_n(t)\) their corresponding time-dependent generalised coordinates. The in-plane quasi-static motion is given by

\[ w_q = \delta \cos(\theta) \left( \frac{x}{\ell} \right) - \frac{\gamma E_q \ell \delta \sin(\theta)}{2\sigma_s^2} \left[ \left( \frac{x}{\ell} \right) - \left( \frac{x}{\ell} \right)^2 \right], \]
Figure 2: Apparatus for inclined cable experiment: a) side view, b) looking along cable from bottom, c) hydraulic actuator and load cell. The cable is 5.4 m long, has diameter 0.00078 m and is inclined at 22.6°. The lead masses are applied to increase the cable mass for more realistic scaling. Deck excitation is simulated with the hydraulic actuator positioned at the lower attachment point.
where the second term on the right-hand side is due to the change in the tension in the cable affecting the static sag of the cable [15, 16]. Furthermore, $\sigma_s$ is the static stress acting along the x-axis and

$$E_q = \frac{1}{1 + \lambda^2/12} E, \quad \text{with} \quad \lambda^2 = \frac{E}{\sigma_s} \left( \frac{\gamma \ell}{\sigma_s} \right)^2, \quad \gamma = \rho g \cos \theta, \quad (3)$$

where $\lambda^2$ is Irvine’s parameter [13]. The system may be linearized assuming that the sag is small compared to the length of the cable, the dynamics along the cable are insignificant and the amplitude of vibration is small compared with the sag [21]. By using the modal decomposition as the mode shapes of this linearized system, the Galerkin technique can be used to derive the modal equations of motion for the nonlinear cable dynamics; here the assumption that the amplitude of vibration is small compared to the sag is relaxed by using a nonlinear compatibility expression. The resulting modal representation of the out-of-plane cable motion for the $n^{th}$ mode may be expressed as

$$m_{yn} \left( \ddot{y}_n + 2\xi_{yn} \omega_{yn} \dot{y}_n + \omega_{yn}^2 y_n \right) + \sum_k \nu_{nk} y_n \left( y_k^2 + z_k^2 \right) + \sum_k 2\beta_{nk} y_n z_k + 2\eta_n \sin(\theta) \delta y_n = 0, \quad (4)$$

and the in-plane cable motion as

$$m_{zn} \left( \ddot{z}_n + 2\xi_{zn} \omega_{zn} \dot{z}_n + \omega_{zn}^2 z_n \right) + \sum_k \nu_{nk} z_n \left( y_k^2 + z_k^2 \right) + \sum_k 2\beta_{nk} z_n z_k + \sum_k \beta_{kn} \left( y_k^2 + z_k^2 \right) + 2\eta_n \sin(\theta) \delta z_n + \zeta_n (-1)^{n+1} \cos(\theta) \ddot{\delta} - \alpha_n \sin(\theta) \dot{\delta} = 0. \quad (5)$$

In these equations $m_{yn} = m_{zn} = m = \rho A \ell/2$ is the effective mass and the parameters $\nu_{nk}$, $\beta_{nk}$, $\eta_n$, $\zeta_n$ and $\alpha_n$ are given by

$$\nu_{nk} = \frac{EA\pi^2 n^2 k^2}{8\ell^3}, \quad \beta_{nk} = \frac{EA\pi n^2}{4\ell \sigma_s} \left( 1 + (-1)^{k+1} \right), \quad \eta_n = \frac{E_q A\pi^2 n^2}{4\ell^2}, \quad \zeta_n = \frac{2m}{n\pi}, \quad \alpha_n = \frac{2m\gamma(\ell E_q)}{n^3 \pi^3 \sigma_s^2} \left( 1 + (-1)^{n+1} \right). \quad (6)$$
Where \( A \) is the cross section area of the cable. The mode shapes for the out-of-plane and even in-plane modes of the linearized system are given by

\[
\phi_n = \sin \left( \frac{n \pi x}{\ell} \right) \quad \text{for} \quad n = 1, 2, 3, \ldots, \quad (7)
\]

\[
\psi_n = \sin \left( \frac{n \pi x}{\ell} \right) \quad \text{for} \quad n = 2, 4, 6, \ldots. \quad (8)
\]

For the odd in-plane modes the mode shapes are more complex; however, as is discussed in [16], for taut cables they may be approximated to sine functions of the same form as the out-of-plane modes. Finally, the out-of-plane and in-plane natural frequencies, \( \omega_{yn} \) and \( \omega_{zn} \), respectively, are given by

\[
\omega_{yn} = \frac{n \pi}{\ell} \sqrt{\frac{\sigma_s}{\rho}}, \quad \omega_{zn} = \frac{n \pi}{\ell} \sqrt{\frac{\sigma_s}{\rho}} (1 + k_n). \quad (9)
\]

Here \( k_n \) is due to the effect of sag and given by

\[
k_n = \left( \frac{2 \lambda^2}{\pi^4 n^4} \right) \left( 1 + (-1)^{n+1} \right)^2, \quad (10)
\]

assuming a sinusoidal mode shape.

The study presented here concentrates on forcing frequencies close to the second natural frequency \( \Omega \approx \omega_{z2} = \omega_{y2} \). This can trigger the 2 : 1 resonance case in the first in- and out-of-plane modes due to the nonlinear nature of the system. Therefore, these four modes will be considered. Note that \( \omega_{y1} = \omega_{z2}/2 \) and \( \omega_{z2} \) is just slightly larger than half the second natural frequency.

Table 1: Cable natural frequencies.

<table>
<thead>
<tr>
<th></th>
<th>( \omega_{y1} ) [rad/s]</th>
<th>( \omega_{y2} ) [rad/s]</th>
<th>( \omega_{z1} ) [rad/s]</th>
<th>( \omega_{z2} ) [rad/s]</th>
</tr>
</thead>
<tbody>
<tr>
<td><strong>Experimental</strong></td>
<td>3.25 ( \cdot 2\pi )</td>
<td>6.51 ( \cdot 2\pi )</td>
<td>3.34 ( \cdot 2\pi )</td>
<td>6.51 ( \cdot 2\pi )</td>
</tr>
<tr>
<td><strong>Theoretical</strong></td>
<td>3.25 ( \cdot 2\pi )</td>
<td>6.50 ( \cdot 2\pi )</td>
<td>3.33 ( \cdot 2\pi )</td>
<td>6.50 ( \cdot 2\pi )</td>
</tr>
</tbody>
</table>

2.2 Scaling and averaging

The technique of scaling and averaging (see for example [16, 22]) is now applied to the four modal equations of motion in- and out-of-plane for \( n = \ldots \)
1.2. We review here the three steps in this procedure; details can be found in [11, 12].

**Step 1** Each modal equation is written in the standard Lagrange form [22]

$$\ddot{v}_i + \omega_{vi}^2 v_i = \epsilon f_i(v_1, v_2, v_3, v_4, \dot{v}_1, \dot{v}_2, \dot{v}_3, \dot{v}_4, \delta, \ddot{\delta}),$$

(11)

where \(\{v_1, v_2, v_3, v_4\} = \{y_1, z_1, y_2, z_2\}\) and \(\{\omega_{v1}, \omega_{v2}, \omega_{v3}, \omega_{v4}\} = \{\omega_1, \omega_1, 2\omega_1, 2\omega_1\}\) with \(\omega_1 = \omega_{y1}\). In this formulation the small parameter \(\epsilon\) has been introduced, indicating that the damping and nonlinear terms are assumed to be small. Equations for the functions \(f_i\) can be derived by comparing (4) and (5) with (11). Note for the case of \(i = 2\), \(f_2\) contains the frequency detuning term \(-\hat{k}_1 \omega_1^2 z_1\) where, since \(k_1\) is small, \(k_1\) has been expressed as \(k_1 = \epsilon \hat{k}_1\) (recalling that \(\omega_{z1} = \sqrt{\omega_1^2 + k_1^2}\)).

**Step 2** Because the forcing is close to the second natural frequency of the cable, the forcing frequency \(\Omega\) is written as \(\Omega = 2\omega_1(1 + \mu)\) where \(\mu\) is the frequency detuning parameter, which may be expressed as \(\mu = \epsilon \hat{\mu}\) since it is small. Applying the time transform \(\tau = t(1 + \mu)\) allows (11) to be rewritten as

$$\ddot{v}_i' + \omega_{vi}^2 v_i = \epsilon f_i(v_1, v_2, v_3, v_4, v_1', v_2', v_3', v_4', \delta, \ddot{\delta}') + 2\hat{\mu} \omega_{vi}^2 v_i = \epsilon g_i,$$

(12)

where \(\{\}'\) is the derivative with respect to \(\tau\) and the terms of order \(\epsilon^2\) have been ignored.

**Step 3** Making substitutions of the form

$$v_i = v_{ic} \cos(\omega_{vi} \tau) + v_{is} \sin(\omega_{vi} \tau), \quad v'_i = \omega_{vi}[-v_{ic} \sin(\omega_{vi} \tau) + v_{is} \cos(\omega_{vi} \tau)],$$

(13)

into (12) results in equations for the dynamics of the amplitude content of the modes

$$v_{ic}' = -\frac{\epsilon}{\omega_{vi}} \sin(\omega_{vi} \tau) g_i, \quad v_{is}' = \frac{\epsilon}{\omega_{vi}} \cos(\omega_{vi} \tau) g_i$$

(14)

(see [16, 22]). Since (14) indicates that the derivatives of \(v_{ic}\) and \(v_{is}\) are small (of order \(\epsilon^1\)) the equations can be averaged. Specifically, this means that the equations are averaged over a period of oscillation at frequency \(\omega_1\) while treating the \(v_{ic}\) and \(v_{is}\) terms within \(g_i\) as constant over the oscillation. This averaging over a period of oscillation at frequency \(\omega_1\) in the transformed \(\tau\)-time domain is equivalent to averaging over two periods of oscillation at frequency \(\Omega\) in the \(t\)-time domain.
The resulting averaged equations are:

\[
\begin{align*}
    y'_{1c} &= \left( \frac{W_{12}z_{1c}z_{1s}}{16\omega_1} \xi_1 \omega_1 \right) y_{1c} - \left( \mu_\omega \omega - \frac{N_1 \Delta}{4\omega_1} - \frac{W_{12}}{32\omega_1} \left( C_1 - 2z_{1c}^2 \right) \right) y_{1s}, \\
    y'_{1s} &= \left( \mu_\omega - \frac{N_1 \Delta}{4\omega_1} - \frac{W_{12}}{32\omega_1} \left( C_1 - 2z_{1s}^2 \right) \right) y_{1c} - \left( \xi_\omega + \frac{W_{12}z_{1c}z_{1s}}{16\omega_1} \right) y_{1s}, \\
    y'_{2c} &= \left( \frac{W_{12}z_{2c}z_{2s}}{2\omega_1} - 2\xi_\omega \right) y_{2c} - \left( 2\mu_\omega - \frac{W_{12}}{8\omega_1} \left( C_2 - 4z_{2c}^2 \right) \right) y_{2s}, \\
    y'_{2s} &= \left( 2\mu_\omega - \frac{W_{12}}{8\omega_1} \left( C_2 - 4z_{2s}^2 \right) \right) y_{2c} - \left( 2\xi_\omega + \frac{W_{12}z_{2c}z_{2s}}{2\omega_1} \right) y_{2s}, \\
    z'_{1c} &= \left( \frac{W_{12}y_{1c}y_{1s}}{16\omega_1} - \xi_\omega \right) z_{1c} - \left( (\mu - \kappa)\omega_1 - \frac{N_1 \Delta}{4\omega_1} - \frac{W_{12}}{32\omega_1} \left( C_1 - 2y_{1c}^2 \right) \right) z_{1s}, \\
    z'_{1s} &= \left( (\mu - \kappa)\omega_1 - \frac{N_1 \Delta}{4\omega_1} - \frac{W_{12}}{32\omega_1} \left( C_1 - 2y_{1s}^2 \right) \right) z_{1c} - \left( \xi_\omega + \frac{W_{12}y_{1c}y_{1s}}{16\omega_1} \right) z_{1s}, \\
    z'_{2c} &= \left( \frac{W_{12}y_{2c}y_{2s}}{2\omega_1} - 2\xi_\omega \right) z_{2c} - \left( 2\mu_\omega - \frac{W_{12}}{8\omega_1} \left( C_2 - 4y_{2c}^2 \right) \right) z_{2s}, \\
    z'_{2s} &= \left( 2\mu_\omega - \frac{W_{12}}{8\omega_1} \left( C_2 - 4y_{2s}^2 \right) \right) z_{2c} - \left( 2\xi_\omega + \frac{W_{12}y_{2c}y_{2s}}{2\omega_1} \right) z_{2s} - B \Delta \omega_1,
\end{align*}
\]

where \( M_1 = y_{1c}^2 + y_{1s}^2 + z_{1c}^2 + z_{1s}^2, M_2 = y_{2c}^2 + y_{2s}^2 + z_{2c}^2 + z_{2s}^2, C_1 = 3M_1 + 8M_2 \) and \( C_2 = M_1 + 6M_2 \). Furthermore, \( \kappa = k_1/2, W_{nk} = \nu_{nk}/m, N_n = 2\eta_n \sin \theta/m, B = \dot{\xi}_2 \cos \theta/m \) and \( \delta = \Delta \cos(\Omega t) \).

Table 2: Cable parameters; note that \( B, \xi \) and \( \kappa \) are nondimensional.

<table>
<thead>
<tr>
<th>( N_1 ) [Hz²/m]</th>
<th>( W_{12} ) [1/(s.m)²]</th>
<th>( \omega_1 ) [rad/s]</th>
<th>( B )</th>
<th>( \xi )</th>
<th>( \kappa )</th>
</tr>
</thead>
<tbody>
<tr>
<td>1.04×10⁻⁴</td>
<td>5.19×10⁻⁴</td>
<td>20.4852</td>
<td>0.2939</td>
<td>0.002</td>
<td>0.0234</td>
</tr>
</tbody>
</table>
3 Experimental cable set-up

Having defined the analytical model, we now introduce the physical cable experiment that was used for this study [23, 24]. The inclined cable used in the experimental tests for this study is shown in Fig. 2. It is 5.4 m long, has diameter 0.00078 m and is inclined at 22.6°. To improve the scaling characteristics of the cable, 22 lead masses have been added to increase its mass. The masses are spaced every 250 mm, except the first and the last ones which have distance 200 mm from each end; Fig. 2(b). The masses are also used in the acquisition of data during the experimental tests, namely as targets that the high-speed camera system follows. Specifically, two masses — one in the middle and the other a quarter distance along the cable — are used to measure the vibration of the cable. This data is acquired using an Imetrum Video Gauge System (VGS) consisting of two cameras, one to record in-plane motion and the second to capture the out-of-plane motion.

The cable is fixed at the top where it is possible to adjust the static tension prior to each test, and all results shown here are for a value of $T_s=286$ N. At the lower end the cable is connected to a short steel beam via a multiaxial (six DoF) load cell and a vertical LVDT with limit displacement of 10mm. The actuator, shown in Fig. 2(c), has 10 kN maximum force and 150 mm displacement. The hydraulic system uses an oil pump with 100 l/min capacity and pressure of 23 MPa.

The system used for the input and output data involves three computers. Computer 1 is used to control the hydraulic actuator, and this is where the required forcing amplitude and frequency for different kinds of tests can be specified. Computer 2 is used for data acquisition of all instruments except the VGS. The third computer is used to record the motion of the cable from the VGS.

During a test, the actuator imitates the deck motion and excites the cable with a sine wave input. Data is recorded using the VGS, load cell and LVDT. The data is post-processed to compute the steady-state amplitudes for the mid- and quarter-point cable displacements in both the in- and out-of-plane directions. Two important quantities are measured in the experiment: (i) the forcing amplitude value at which any out-of-plane motion begins, and (ii) the steady state maximum amplitude per forcing period. A discussion of how this experimental data compares with a bifurcation analysis of the scaled, averaged model is given in the next section.
4 Bifurcation study

The connection between the solutions of system (15) and the vibration response of the cable is given by the amplitude contributions of the four different modal amplitudes, which are defined as:

- \( Z_2 = \sqrt{z_{2c}^2 + z_{2s}^2} \) for the second in-plane modal amplitude;
- \( Z_1 = \sqrt{z_{1c}^2 + z_{1s}^2} \) for the first in-plane modal amplitude;
- \( Y_2 = \sqrt{y_{2c}^2 + y_{2s}^2} \) for the second out-of-plane modal amplitude;
- \( Y_1 = \sqrt{y_{1c}^2 + y_{1s}^2} \) for the first out-of-plane modal amplitude.

The main parameters of interest of system (15) are the amplitude of excitation \( \Delta \) (measured in meters) and the detuning \( \mu \) between the frequency of the actuator and the second natural frequency of the cable. In order to work with non-dimensional values for the actuator amplitude, we consider throughout the normalized quantity \( \Delta/L \), where \( L \) is the untensioned length of the cable which was \( L = 5.4 \) m in the experiment, here \( L = l \) the support separation distance.

Our goal now is to investigate how the contributions of the different modal amplitudes \( Z_2, Z_1, Y_2 \) and \( Y_1 \) of equations (15) change as a function of \( \Delta \) and \( \mu \). To this end, we present in Figs. 3 and 4 a bifurcation diagram in the \((\mu, \Delta/L)\)-plane that explains how the different modal amplitudes contribute to the overall behavior. The curves in Figs. 3 and 4 represent transitions that are due to changes of stability of individual modal amplitudes; they have been computed with the numerical continuation package AUTO [18]. These transitions are known bifurcations [25, 26], and we find two types of bifurcations in the region of interest \((\mu, \Delta/L) \in [0, 0.0015] \times [-0.03, 0.07]\). The first is the saddle-node (or fold) bifurcation, where a stable and an unstable branch of solutions meet; secondly, we find branch point bifurcations that correspond to the onset of the contribution of a further modal amplitude to the solution.

Figure 3 shows loci of fold bifurcation points of the \( Z_2 \)-amplitude as the grey curve labelled \( F_{Z_2} \), as well as black curves \( B_{Y_1} \) and \( B_{Z_1} \) of branch point bifurcations of the \( Y_1 \)-amplitude and of the \( Z_1 \)-amplitude, respectively. These bifurcations are observable because they correspond to stable solutions. In the lower region (below these curves) the stable solution of Equations (15)
Figure 3: Partial bifurcation diagram of Equations (15) in the \((\mu, \Delta/L)\)-plane showing a curve \(F_{Z_2}\) of fold bifurcations of \(Z_2\) and curves \(B_{Z_1}\) and \(B_{Y_1}\) of branch points of \(Z_1\) and \(Y_1\), respectively. Also shown are experimental data points; here □ indicates the largest \(\Delta/L\)-value where only \(Z_2\) was detected, △ where \(Y_1\) was detected for the first time, ◇ where \(Z_1\) was detected for the first time, and ▽ that \(Y_2\) was detected.

has only a contribution of the \(Z_2\)-amplitude (that is, \(Z_1 = Y_2 = Y_1 = 0\)), meaning that the response of the cable consists of purely the second in-plane mode. When the curve \(F_{Z_2}\) (between the points C and K) is crossed as \(\Delta/L\) is increased, we encounter the fold bifurcation and this stable solution disappears. As a consequence, there is a sudden transition to an entirely different stable solution that has contributions of the \(Z_2\)-amplitude and both the \(Y_2\)-amplitude and the \(Y_1\)-amplitude, meaning that the cable now features out-of-plane motion; how this new solution arises is discussed in Sec. 4.1 below. Crossing a branch point bifurcation curve in Fig. 3 has less dramatic consequences. When the curve \(B_{Z_1}\) is crossed for increasing \(\Delta/L\) one finds an onset of the contribution of the \(Z_1\)-amplitude, while still \(Y_2 = Y_1 = 0\); hence, the cable dynamics is still in-plane. Similarly, when the curve \(B_{Y_1}\) is crossed for increasing \(\Delta/L\) the \(Y_1\)-amplitude starts to contribute, while still \(Z_1 = Y_2 = 0\); in other words, one observes the onset of the first out-of-plane
modal amplitude.

The bifurcation curves shown in Fig. 3 are the transition curves that one encounters first in the experiment when $\Delta/L$ is increased for fixed detuning $\mu$. Indeed, these curves agree well with the experimental measurements. The squares are the last measured points where $Z_1 = Y_2 = Y_1 = 0$, and the triangle where a contribution $Y_1 \neq 0$ of the first out-of-plane $Y_1$-amplitude was identified for the first time (when increasing $\Delta/L$). Similarly, the diamonds indicate that a nonzero contribution of the first in-plane $Z_1$-amplitude was first observed. Given the gradual onset of the corresponding modal amplitudes, these measurements points agree well with the curves $B_{Z_1}$ and $B_{Y_1}$, respectively.

How the curves shown in Fig. 3 arise can only be understood when they are considered as part of an overall bifurcation diagram that also shows bifurcation curves that correspond to unstable solutions. This is shown in Fig. 4, where the curves $F_{Z_2}$, $B_{Z_1}$, $B_{Y_2}$ and $B_{Y_1}$ are seen to form a complicated structure. Throughout, parts of curves that correspond to bifurcation of stable solutions are drawn as solid curves, while parts of curves that correspond to bifurcations of unstable solutions are dashed. Notice the solid curves from Fig. 3 are parts of the corresponding bifurcation curves. More specifically, the fold curve $F_{Z_2}$ has a cusp point $C$ and extends past the point $K$, where it meets the curve $B_{Z_1}$; apart from the part of $F_{Z_2}$ between $C$ and $K$ that was shown in Fig. 3. Note also the part of curve $F_{Z_2}$ between $C$ and the point $P$ corresponds to the bifurcation of a stable solution. However, this fold bifurcation is almost immediately followed by the curve $B_{Y_2}$ of branch point bifurcations of the second out-of-plane $Y_2$-amplitude.

Notice that the curve $B_{Y_2}$ is solid and corresponds to bifurcations of an attracting solution; see the enlargement in Fig. 4(b) showing $(\mu, \Delta/L) \in [0, 0.00011] \times [-0.03, 0.07]$. Hence, the region of stability of the stable solution in between $F_{Z_2}$ from $C$ to $P$ and the solid part of $B_{Y_2}$ is extremely narrow, and as a result it is very difficult to observe a stable solution with a contribution of only the $Y_2$-amplitude; this is why these solid parts of $F_{Z_2}$ and $B_{Y_2}$ are not shown in Fig. 3. Nevertheless, we detected the onset of the $Y_2$-amplitude in an experiment for $\mu = 0.02$; it is represented by the upside-down triangle. Notice further that the curves $B_{Y_1}$ and $B_{Z_1}$ extend past the points $N$ and $K$, respectively, where they indicate that the corresponding modal amplitude starts to contribute to an unstable solution.

At the points $N$, $K$ and $P$ two different bifurcation curves become tangent, which indicates a bifurcation of codimension two [26]. As a result the
Figure 4: Complete bifurcation diagram of Equations (15) in the $(\mu, \Delta/L)$-plane showing a curve $FZ_2$ of fold bifurcations and curves $BZ_1$, $BY_1$ and $BY_2$ of branch point bifurcations; along solid parts of curves the bifurcation concerns a stable solution and along dashed parts it concerns an unstable solution. The experimental data points are repeated from Figure 3; panel b) is an enlargement for small $\Delta/L$.  

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nature of at least one of the bifurcation curves involved changes from involving a stable to involving an unstable solution. Overall, the bifurcation curves $F_{Z_2}$, $B_{Z_1}$, $B_{Y_2}$ and $B_{Y_1}$ form a consistent bifurcation diagram in the $(\mu, \Delta/L)$-plane that explains the nature of the overall solution structure. We finally remark that this solution structure can be represented by surfaces of solutions over the $(\mu, \Delta/L)$-plane, when an appropriate norm is used to represent the respective solutions; this is beyond the scope of this paper and will be reported elsewhere.

### 4.1 Amplitudes of vibration

The bifurcation diagram in the $(\mu, \Delta/L)$-plane shown in Fig. 4 features quite a number of bifurcation curves in the region of positive $\mu$. In fact, some of these bifurcations give rise to additional branches of stable solutions, which are connected to the known stable solutions discussed above by branches of unstable solutions. To show this, Fig. 5 shows the one-parameter bifurcation diagram of Equations (15) for fixed $\mu = 0.02$; here solution branches, computed by numerical continuation in $\Delta/L$, are represented by the norm $||N|| = \sqrt{Z_1^2 + Z_2^2 + Y_1^2 + Y_2^2}/L$ (which measures contributions from all four modal amplitudes of the cable). Stable parts of solution branches are shown as black curves, and unstable parts as grey curves. The continuation started for $\Delta/L$ by following the solution curve $l_1$, where only the second in-plane amplitude $Z_2$ has a contribution to the vibration response of the cable. At $\Delta/L$ approximately $1.56 \times 10^{-4}$ a branch point $B_1$ to another solution branch $l_2$ is detected; beyond this point the solution curve $l_1$ becomes unstable, while the branch $l_2$ is stable, where the first in-plane amplitude $Z_1$ has a contribution as well as $Z_2$. At $\Delta/L$ approximately $1.8781 \times 10^{-4}$ there is branch point $B_2$ where $l_2$ and $l_1$ meet and branch $l_1$ regains stability until the fold point $F_1$ is reached.

In terms of the dynamics of the cable, moving $\Delta/L$ beyond $2.0372 \times 10^{-4}$ there is a jump to another stable solution branch, namely the branch $l_3$ where we find responses from all the modal amplitudes considered except $Z_1$. At $\Delta/L$ approximately $3.9989 \times 10^{-4}$ a branch point $B_3$ is detected where $l_3$ meets the solution branch $l_4$, along which the first out-of-plane modal amplitude $Y_1$ is coupled with $Z_2$. This branch is stable until the branch leaves the $\Delta/L$ range of interest in Fig. 5. Note that the stable branches $l_3$ and $l_4$ are connected to the stable branches $l_1$ and $l_2$ via branches of unstable solutions. More specifically, there is an unstable part of $l_1$ that
Figure 5: One-parameter bifurcation diagram of Equation (15) for $\mu = 0.02$, showing solution branches represented by their norm $||N|| = \sqrt{Z_1^2 + Z_2^2 + Y_1^2 + Y_2^2}/L$ for $\Delta/L \in [0, 6 \times 10^{-4}]$. The stable solution branches $l_1$–$l_4$ (black curves) are connected via unstable solution branches (grey curves), which meet at bifurcation points; specifically, The $B_1$, $B_2$, and $B_3$ are branch points (denoted by ■), $F_1$ and $F_2$ are fold points (denoted by •), and $H$ is a Hopf bifurcation point (denoted by *).

extends beyond the fold point $F_1$ and past a second fold point $F_2$. Near $F_2$ one finds a branch point bifurcation to an unstable branch where the modal amplitudes $Y_2$ and $Z_2$ contribute. Finally, the unstable part of $l_4$, to the left of the branch point $B_3$ extends past a fold bifurcation and connects to the former branch. The fact that all of these branches are connected allowed us to continue them (irrespective of their stability), yielding an overall and consistent one-parameter bifurcation diagram. It is a particular strength of the continuation approach that new branches of (stable) solutions can be found in a systematic fashion.

In order to facilitate the physical interpretation of the stable branches, Fig. 6 shows the contributions of $Z_2$, $Z_1$, $Y_2$, and $Y_1$ modal amplitudes to the stable branches $l_1$–$l_4$ in separate panels. This allows us to see clearly how much a particular modal amplitude contributes to the cable steady-state
Figure 6: Mode contributions to the stable branches $l_1$-$l_4$ from Figure 5 ($\mu = 0.02$), showing the contribution of a) $Z_2$, b) $Z_1$, c) $Y_2$, and d) $Y_1$. 

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vibration response. As Fig. 6(a) shows, the modal amplitude $Z_2$ contributes to all stable branches; as $\Delta/L$ grows, the response of the cable in $Z_2$ grows until the fold point $F_1$ is reached. The contributions of all other modal amplitudes are zero, except for a small contribution of $Z_1$ along the branch $l_2$; see Fig. 6(b). As was mentioned before, when $\Delta/L$ is increased beyond $F_1$, the only available stable solutions is that on branch $l_3$, which features strong contributions from $Y_2$ and $Y_1$; see Fig. 6(c) and (d). The contribution of $Y_2$ then vanishes when the branch point $B_3$ is reached, and along the branch $l_4$ only the modal amplitudes $Z_2$ and $Y_1$ are active. Notice that at the frequency considered $\mu = 0.02$, except along the small branch $l_2$, the modal amplitude $Z_1$ does not contribute to the stable dynamics of the cable.

4.2 Comparison between analysis and experiment

In the experiments we measure the amplitudes of the in-plane displacement $w$ and the out-of-plane displacement $v$ at the mid-point and the quarter point of the cable, we define their maximums with $W$ and $V$ respectively. To allow for a comparison between our bifurcation analysis and experimental data we show in Fig. 7 the stable branches in terms of the maximal displacements as given by

\[
\frac{W}{L} \left( \frac{L}{4}, \Delta \right) = \frac{Z_1}{\sqrt{2}L}(\Delta) + \frac{Z_2}{L}(\Delta) + \frac{w_q}{L} \left( \frac{L}{4}, \Delta \right), \tag{16}
\]

\[
\frac{W}{L} \left( \frac{L}{2}, \Delta \right) = \frac{Z_1}{L}(\Delta) + \frac{w_q}{L} \left( \frac{L}{2}, \Delta \right),
\]

\[
\frac{V}{L} \left( \frac{L}{4}, \Delta \right) = \frac{Y_1}{\sqrt{2}L}(\Delta) + \frac{Y_2}{L}(\Delta),
\]

\[
\frac{V}{L} \left( \frac{L}{2}, \Delta \right) = \frac{Y_1}{L}(\Delta), \quad \text{where}
\]

\[
\frac{w_q}{L} \left( \frac{L}{4}, \Delta \right) = 0.2651 \frac{\Delta}{L},
\]

\[
\frac{w_q}{L} \left( \frac{L}{2}, \Delta \right) = 0.1996 \frac{\Delta}{L}.
\]

Here we take into account the contribution of the quasi-static motion $w_q$, (2), produced by the actuator of the cable [12] to the in-plane displacement $w$. Furthermore, because we consider the maximum displacement, low damping,
Figure 7: Comparison of theoretical maximum displacement with experimental measurements for $\mu = 0.02$. Shown are measured data points of displacement (denoted by $\bigcirc$), together with theoretical values along the stable branches $l_1$–$l_4$ of a) $W(\frac{L}{4})/L$, b) $W(\frac{L}{2})/L$, c) $V(\frac{L}{4})/L$, and d) $V(\frac{L}{2})/L$. 

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and no phase lag in response of the cable, in the equation for \( w_q \), we use \( \Delta/L \) instead of \( \delta/L = (\Delta/L) \cos(\Omega t) \).

The in-plane quarter- and mid-point displacements along the stable branches \( l_1 - l_4 \) are shown in Fig. 7(a) and (b), and their out-of-plane quarter- and mid-point displacements in Fig. 7(c) and (d). The white circles are measured experimental data points. For each data point the experiment was started at the rest position of the cable for the associated value of the excitation amplitude \( \Delta/L \). After transients died down, the in-plane and out-of-plane amplitudes at the quarter- and mid-points of the cable were measured; the maximum excitation amplitude that was considered was \( \Delta/L = 2.33 \times 10^{-4} \).

In all panels of Fig. 7 the measured amplitudes agree very well with the predicted values. In particular, the measurements shows that, up to the fold point \( F_1 \) at \( \Delta/L \approx 2.0372 \times 10^{-4} \), the cable indeed displays practically zero in-plane amplitude at the mid point \( L/2 \), as well as no out-of-plane dynamics.

What is more, beyond the point \( F_1 \) the cable response changes its nature and out-of-plane modal amplitudes start contributing. Indeed, we measured nonzero out-of-plane amplitudes \( V(L/4) \) and \( V(L/2) \), as well as a considerably higher amplitude at \( W(L/4) \). This experimental observation agrees well with the theoretical prediction. However, the prediction of the actual measured amplitudes is less accurate for \( \Delta/L > 2.0372 \times 10^{-4} \), that is, for solutions along the stable branch \( l_3 \). While \( V(L/4) \) is predicted quite accurately, there is some mismatch in the measurements of \( W(L/4) \) and \( V(L/2) \); notice further that there is also a measured amplitude contribution of \( W(L/2) \). We remark that, when out-of-plane dynamics of the response is present as well, determining the displacement at the in-plane and out-of-plane amplitudes at quarter- and mid-points reliably over a long period of time is quite a bit harder. We suspect that this difficulty may arise because of a weak interaction between the out-of-plane \( Y_1 \)-amplitude and quasi-static in-plane motion. Notice further that the experiment needs to be run longer to allow transients to die down, which also introduces an additional level of uncertainty concerning the exact measured amplitudes.

In spite of these technical difficulties, overall Fig. 7 clearly shows that the displacements of the cable, even for \( \Delta/L \) above \( F_1 \), are predicted well by the stable parts of solution branches found by numerical continuation for the averaged system (15) representing the vibration response of the vertically excited cable. More specifically, with the continuation technique it is possible to predict accurately when the response of the cable changes from uncoupled in-plane modal amplitude dynamics to dynamics with a contribution of the
out-of-plane modal amplitudes. Moreover, it is possible to predict from the four-mode model exactly which modal amplitudes are involved in these coupled responses and with which amplitudes.

5 Conclusions

In this paper we have presented a study of the nonlinear vibration of an inclined cable that is excited at its lower attachment point. This type of cable vibration is of interest because inclined cables are used to support the bridge deck in cable-stay bridges. Previous work had identified the lower stability boundaries for out-of-plane, sway motion in terms of the excitation amplitude and frequency, and validated them experimentally. In this paper we extended this work by (i) determining the complete stability diagram in the plane of excitation amplitude versus detuning, (ii) showing how additional stable branches are connected to known solutions via unstable branches, and (iii) deriving quantitative amplitude information for the modal displacements of the cable. These theoretical results are in good agreement with experimental measurements that go further than those in [11, 12] in that they also show the computed modal displacements of the cable. To the authors knowledge, this type of modelling and experimental validation has not previously been achieved for modal amplitudes of an inclined cable. Future work will concentrate on non-sinosoidal forcing of the cable and on feedback effects of the cable dynamics on the bridge deck.

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